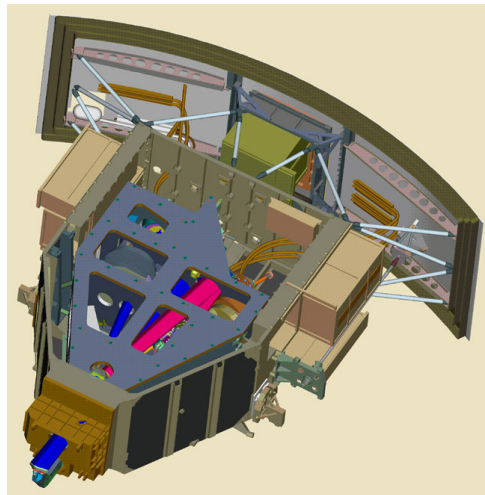
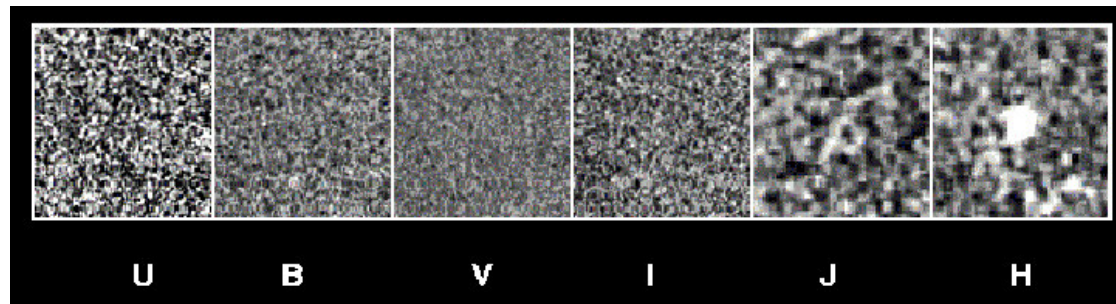
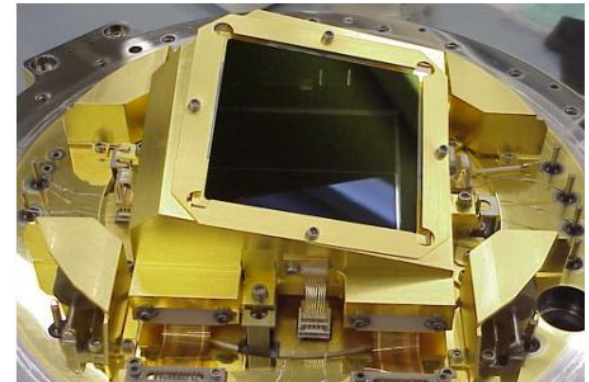


# Wide Field Camera 3

## Update to Origins Subcommittee

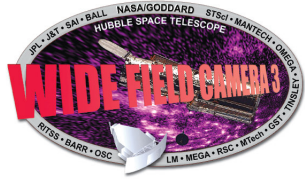


Randy Kimble  
Instrument Scientist  
December 2, 2002



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Wide Field Camera 3 Update to Origins Subcommittee – December 2, 2002



## WFC3 Status Summary (1)



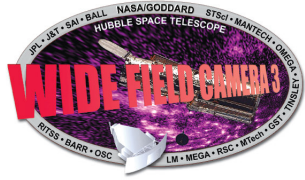
(Why I'm here.) New instrument scientist appointed upon Ed Cheng's departure from GSFC to form private company.

WFC3 continues to make excellent technical progress:

- Optical bench is fully populated with flight optics/filters (except for pick-off mirror) and mechanisms.
- Optical system has been fully aligned w/surrogate detectors and delivers excellent optical performance.
- Optical bench is being prepped for shipment to GSFC on December 9.
- All flight electronics boxes are complete – many have been delivered to Goddard.

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Wide Field Camera 3 Update to Origins Subcommittee – December 2, 2002



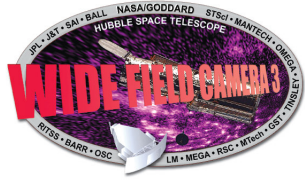
## WFC3 Status Summary (2)



- Modified TECFIRE design is fully qualified and delivers better than required performance for cooling IR channel.
- UVIS flight detector package is complete, incorporating superb Marconi CCDs; now in environmental testing.
- IR detector qual unit successfully completed thermal testing with performance margin.
- IR FPA development has been problematic, but steady progress has been made; *a flyable device is in hand* and additional devices are currently in fabrication.
- Optical Stimulus is through thermal-vac and ready to support I&T at Goddard.
- I&T schedule at GSFC has been reworked to accommodate launch delay in the most cost efficient manner.

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## Scientific Mandate & Capabilities



- Ensure an imaging capability through 2010
  - Provide complementary capabilities to ACS
  - With several years less radiation damage to CCD detectors
- Panchromatic coverage over a wide field
  - 200 to 1700 nm, widest coverage of any HST instrument
- Uniquely capable in the near-UV
  - 200 to 400 nm
  - Higher NUV “discovery efficiency” (throughput  $\times$  FOV) than other HST instruments by 15-30 $\times$
- Uniquely capable in the near-IR
  - 850 to 1700 nm
  - Higher NIR “discovery efficiency” than NICMOS3 by  $>10\times$
- Large and diverse set of filters: 48 UVIS, 16 IR

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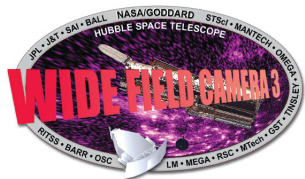




These topics are addressed in detail in the WFC3 Science White Paper.

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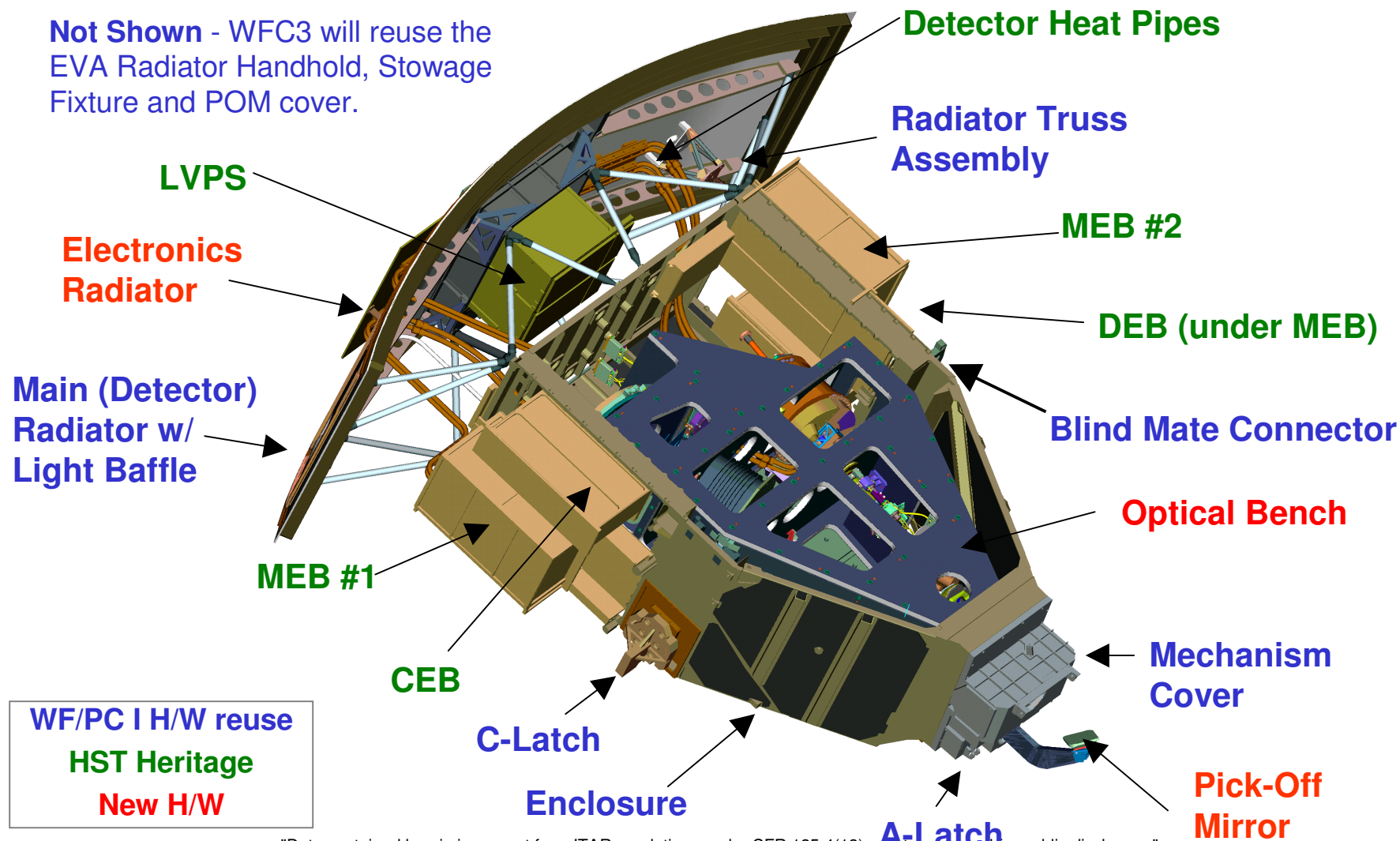




# WFC3 Replaces WFPC2 in Radial Instrument Bay



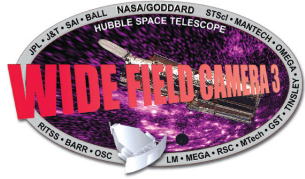
**Not Shown** - WFC3 will reuse the EVA Radiator Handhold, Stowage Fixture and POM cover.



WF/PC I H/W reuse  
HST Heritage  
New H/W

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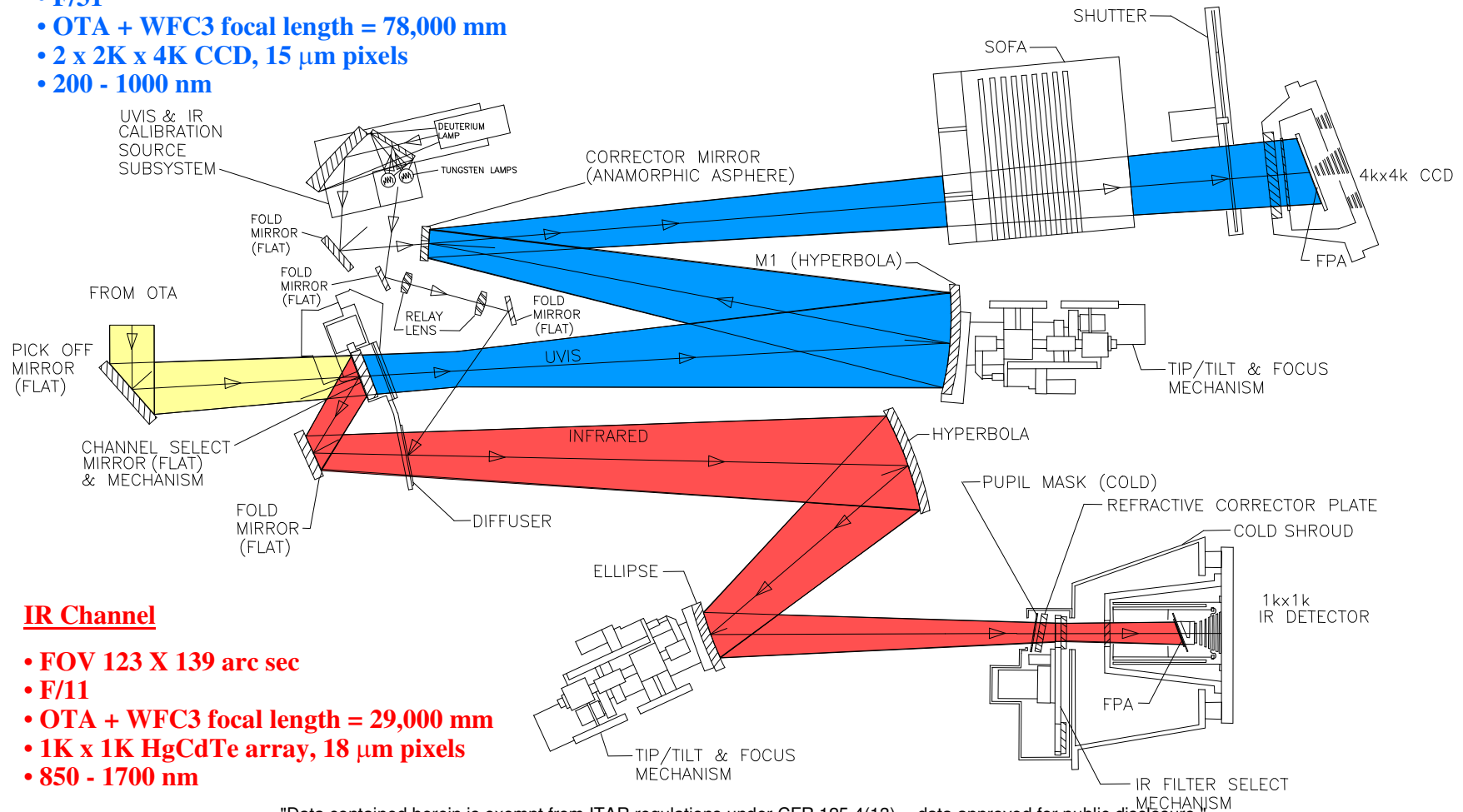


# WFC3 Has Two Optimized Optical Channels



## UVIS Channel

- FOV 160 X 160 arc sec
- F/31
- OTA + WFC3 focal length = 78,000 mm
- 2 x 2K x 4K CCD, 15  $\mu$ m pixels
- 200 - 1000 nm

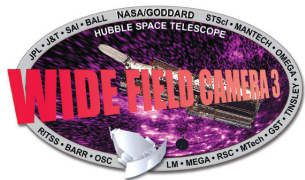


## IR Channel

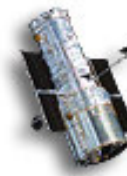
- FOV 123 X 139 arc sec
- F/11
- OTA + WFC3 focal length = 29,000 mm
- 1K x 1K HgCdTe array, 18  $\mu$ m pixels
- 850 - 1700 nm

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# WFC3 Key Performance Parameters



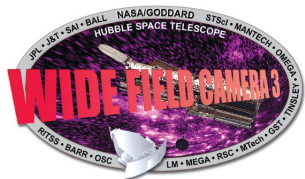
	UVIS	IR	
<b>Format</b>	2 x 2K x 4K	1K x 1K	pixels
<b>Field Size</b>	160 x 160	135 x 135	arcsec
<b>Pixel Size</b>	39	130	mas
<b>Spectral Range</b>	200 to 1000	850 to 1700	nm
<b>Dark Current</b>	< 0.003	< 0.4	e-/pix/sec
<b>Readout Noise</b>	< 4	< 15	e-/pix/ readout
<b>Operating Temp</b>	-90	-120	°C

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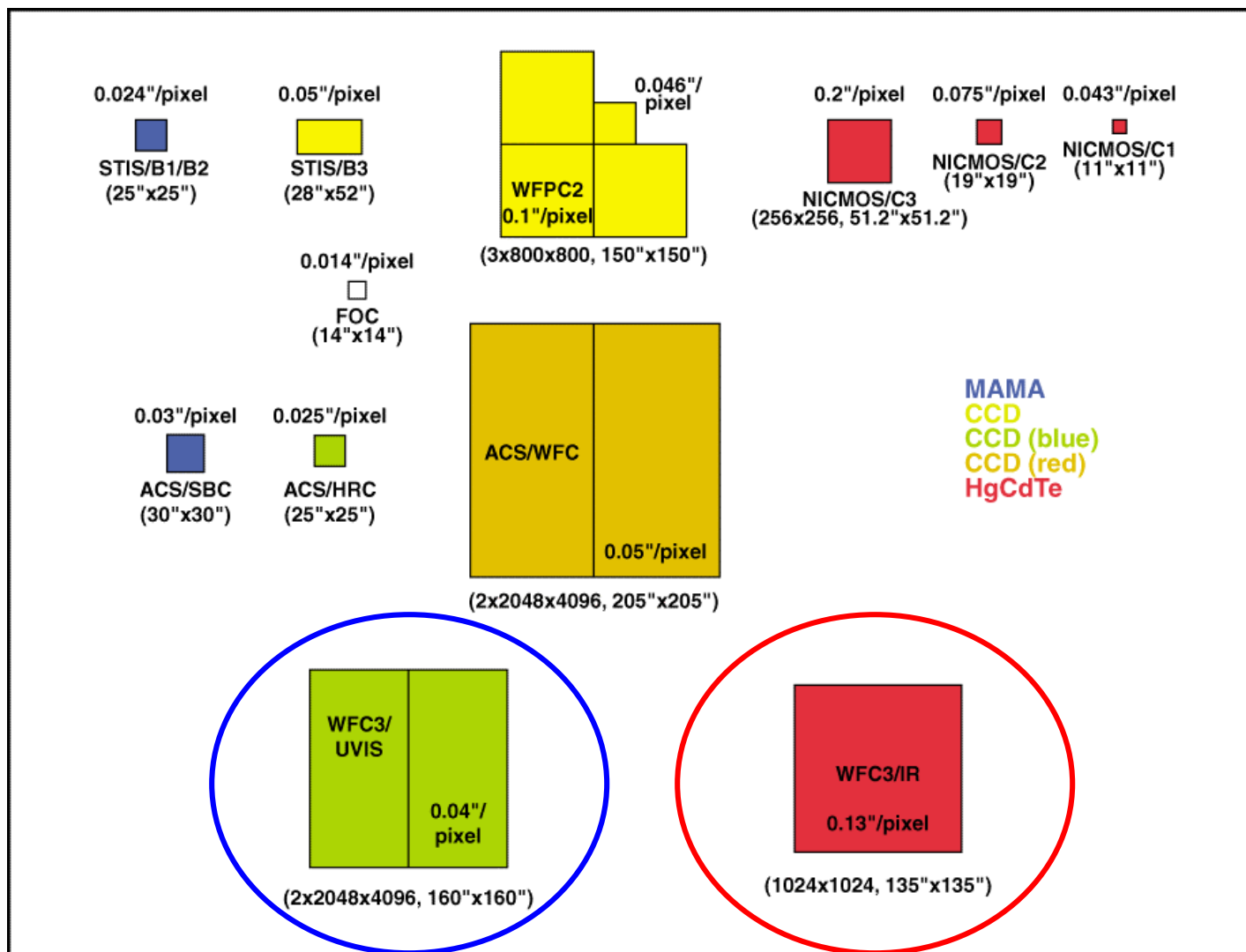
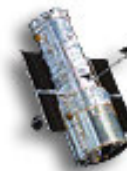
Randy.A.Kimble@nasa.gov, NASA/GSFC Code 681, 301-286-5783  
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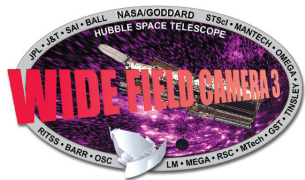


# WFC3 Advances HST Imaging FOV

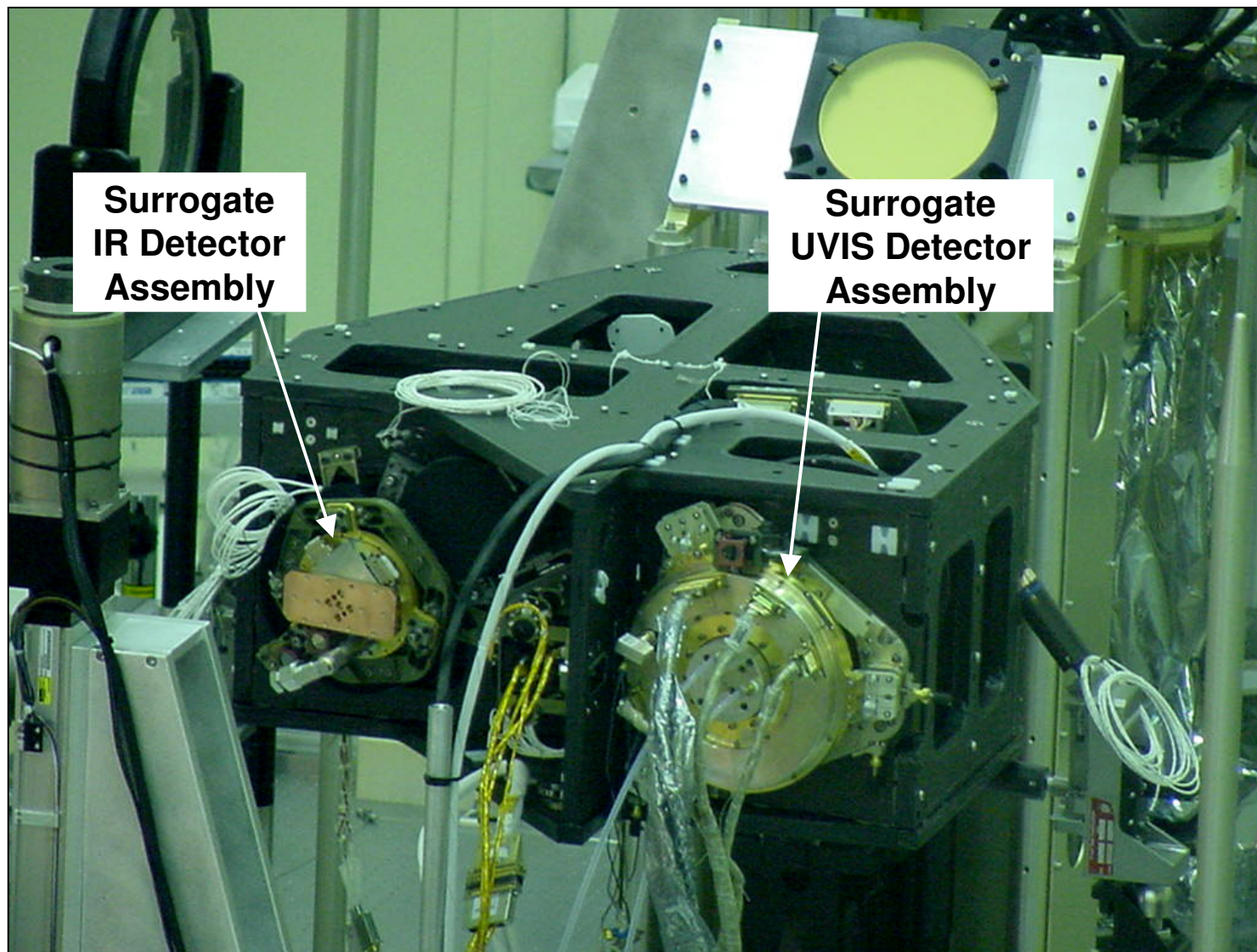


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Randy.A.Kimble@nasa.gov, NASA/GSFC Code 681, 301-286-5783  
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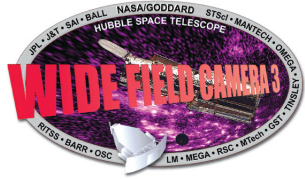


# Optical Bench is Fully Populated and Aligned

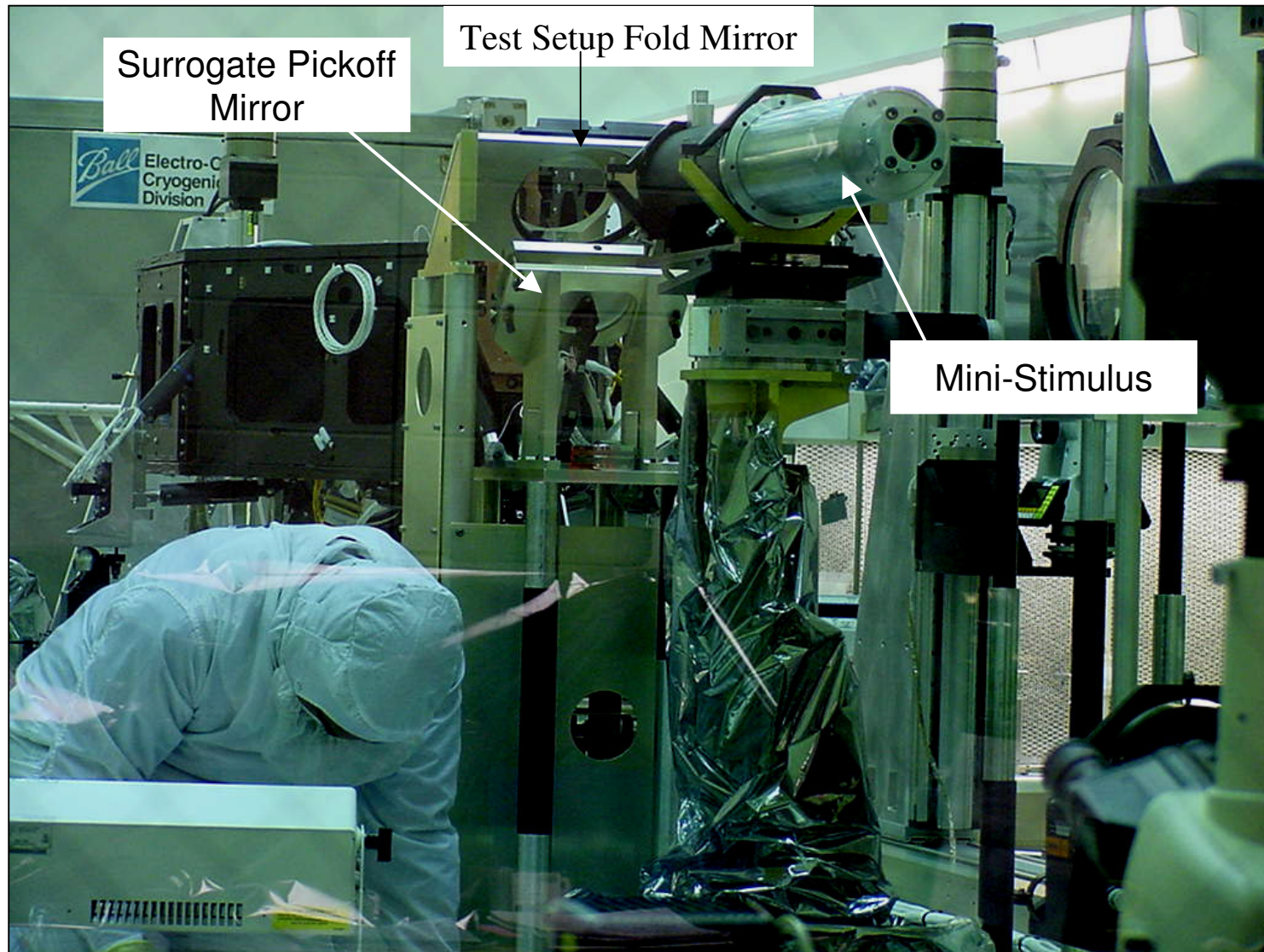


Randy.A.Kimble@nasa.gov, NASA/GSFC Code 681, 301-286-5783  
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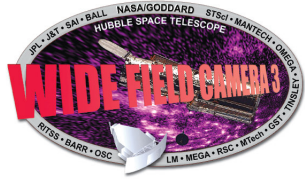




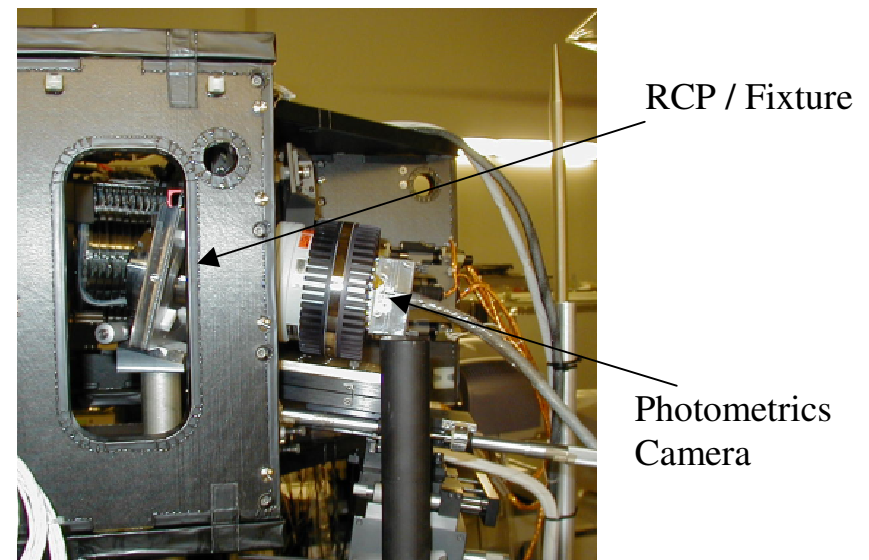
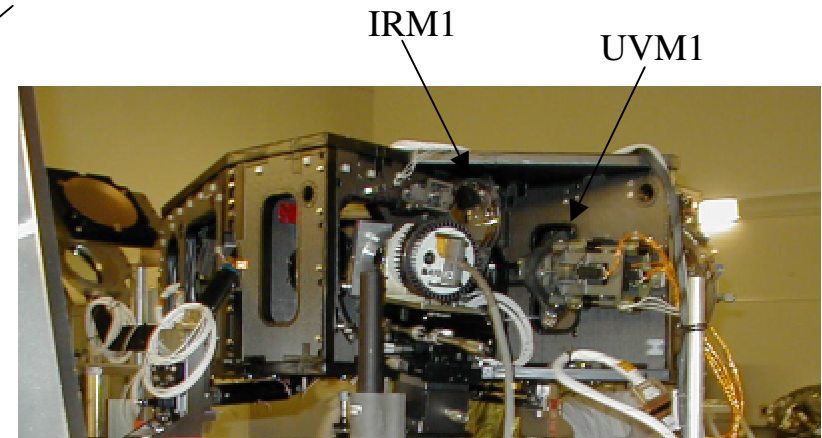
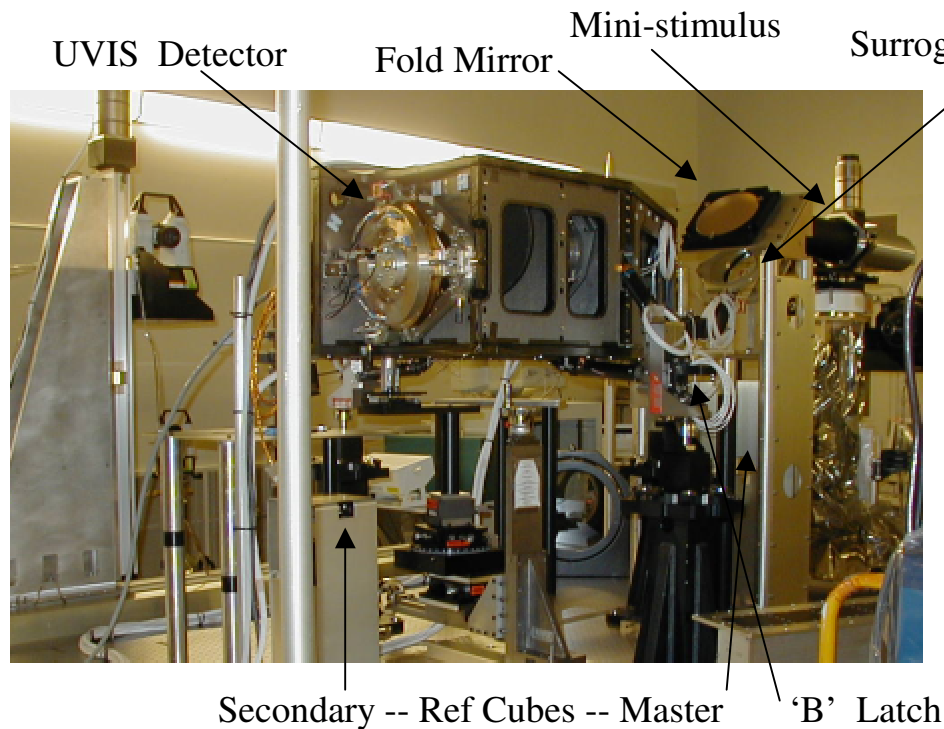
# “Mini-Stimulus” Source Simulates Hubble’s Spherical Aberration



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Wide Field Camera 3 Update to Origins Subcommittee – December 2, 2002



# Mini-stim, WFAS & WFC3

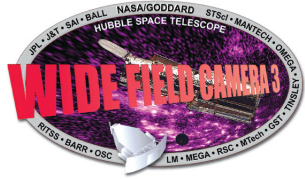


Optical set-up and metrology are complex and precision activities with many data/reference transfers

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# UVIS Encircled Energy Is Better Than Specification



750 WFC3/UVIS AmpD

Field center; optimized

X center: 121.280

Y center: 96.6300

Max value: 29642.2

Min value: -26.8513

Total flux: 144368.57

Background: 0.794318

Peak/total: 0.2057

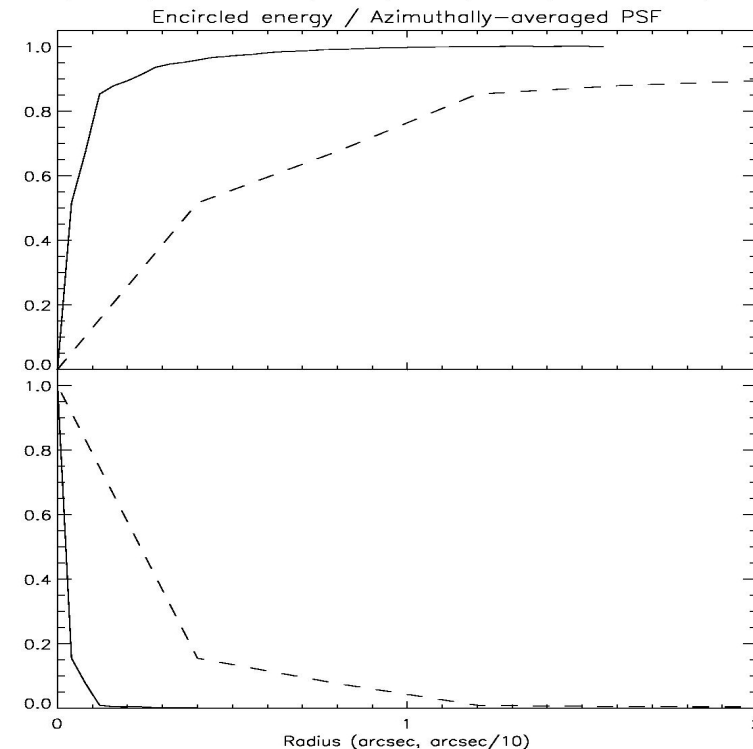
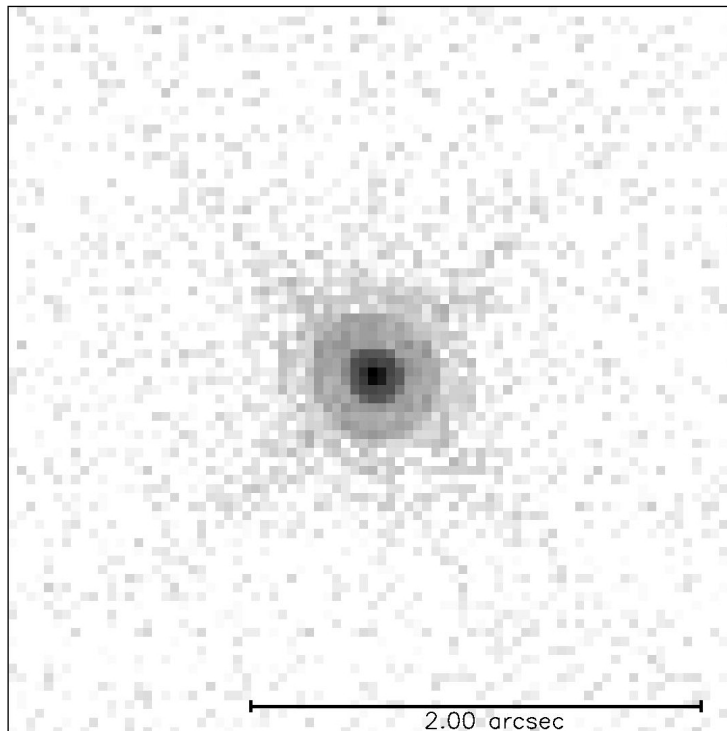
Sharpness: 0.0833

EE(d=.25): 0.8475

EE(d=.40): 0.8948

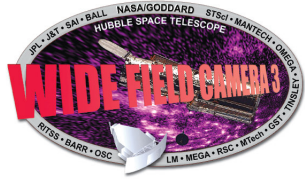


Spec EE at 633 nm  
(d=.25) = 75%



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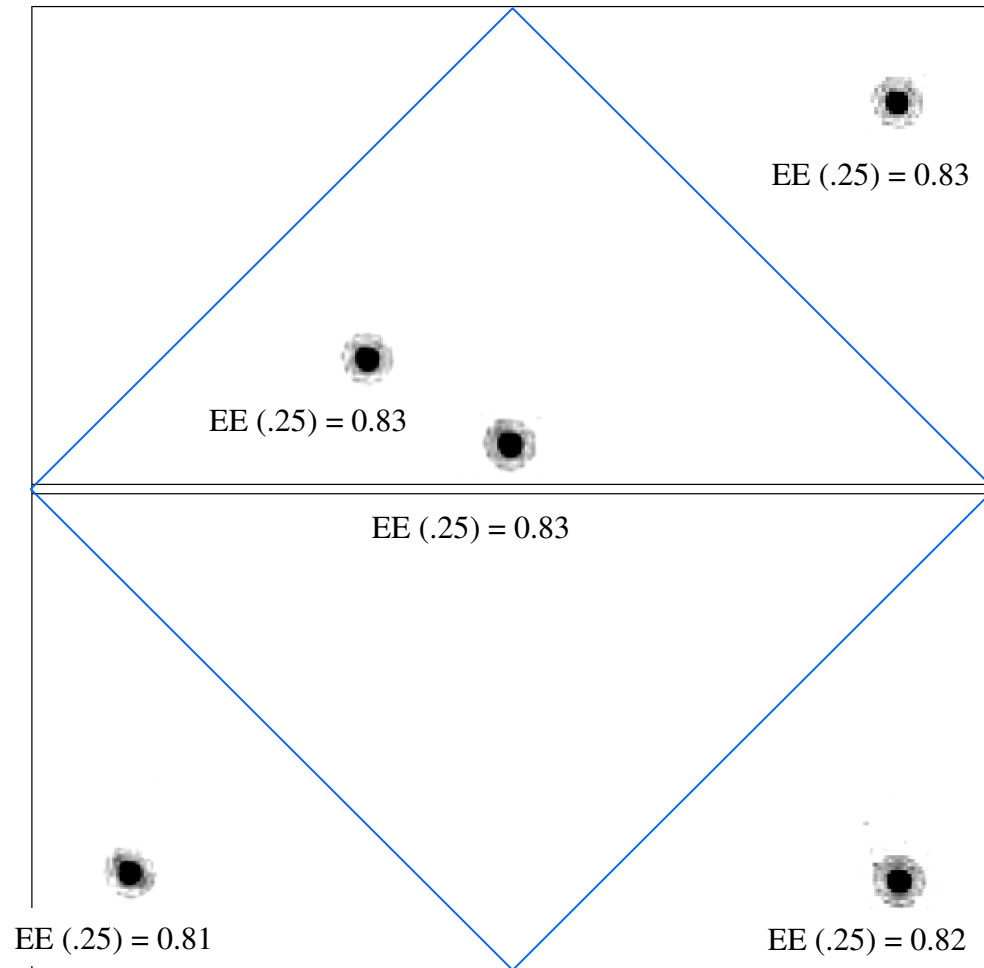
Randy.A.Kimble@nasa.gov, NASA/GSFC Code 681, 301-286-5783  
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# UVIS EE Good Over Field



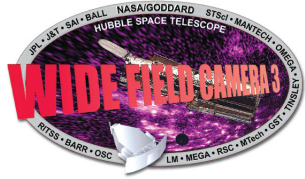
WFC3/UVIS Image Quality Evaluation MiniStim 22 Oct 02



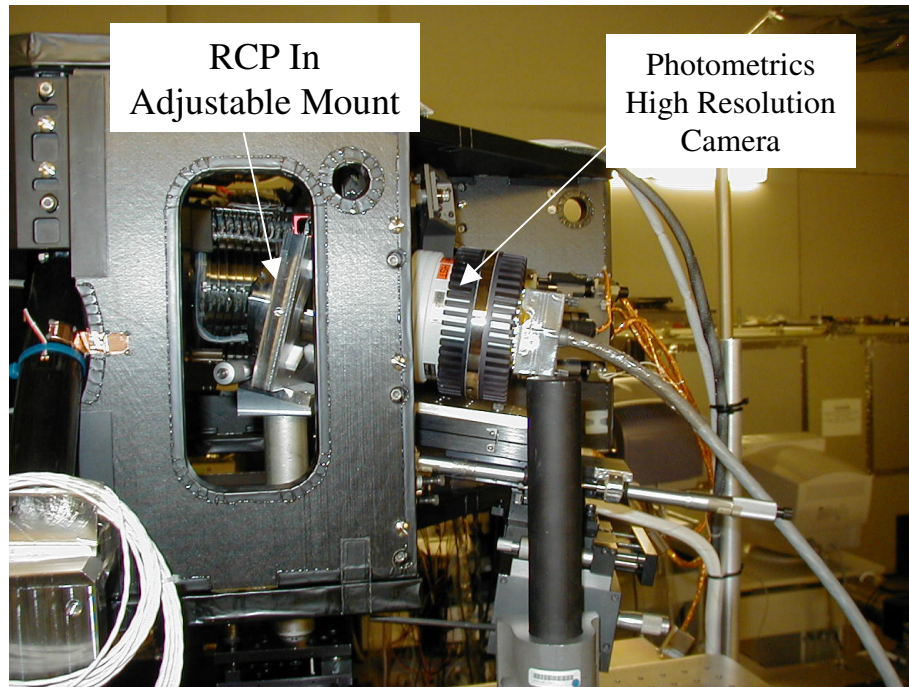
Spec EE at 633 nm  
(d=0.25) = 75%

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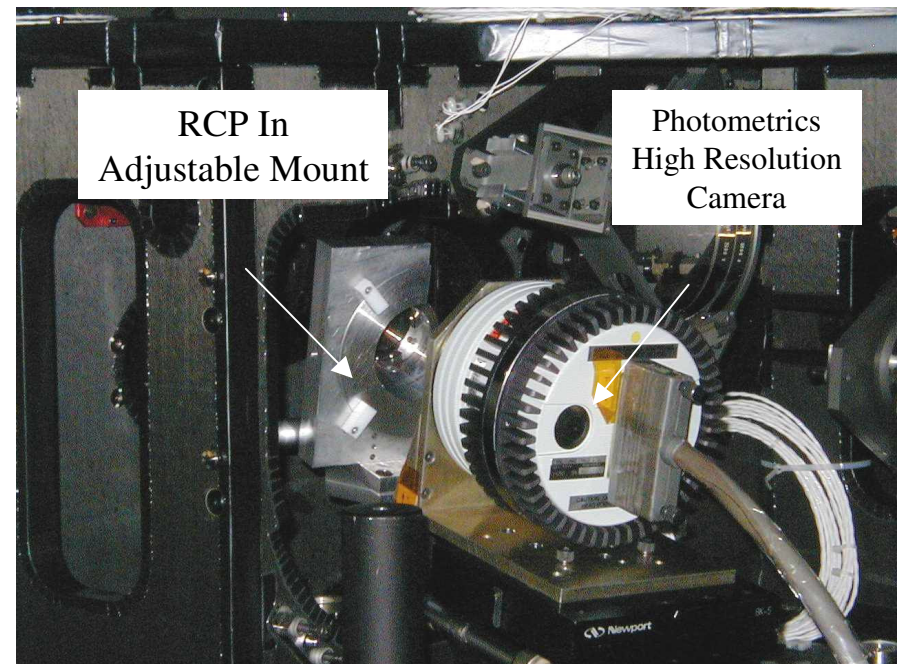


# IR Refractive Corrector Alignment was Difficult but Ultimately Successful



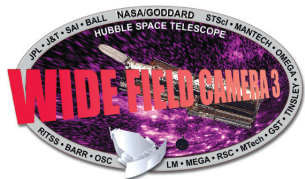
Side view showing RCP installed in 6 degree of freedom adjustable mount with Photometrics high resolution camera installed in place of IR surrogate detector

View from back of optical bench

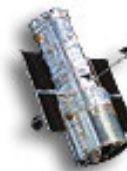


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# IR Channel Encircled Energy at 633 nm Measured with Small-Pixel CCD



1012\_17.fits WFC3/IRPM

Field center; optimized

X center: 668.087

Y center: 415.497

Max value: 3460

Min value: 23

Total flux: 24599.911

Background: 50.4349

Peak/total: 0.1390

Sharpness: 0.0518

EE(d=.25): 0.7367

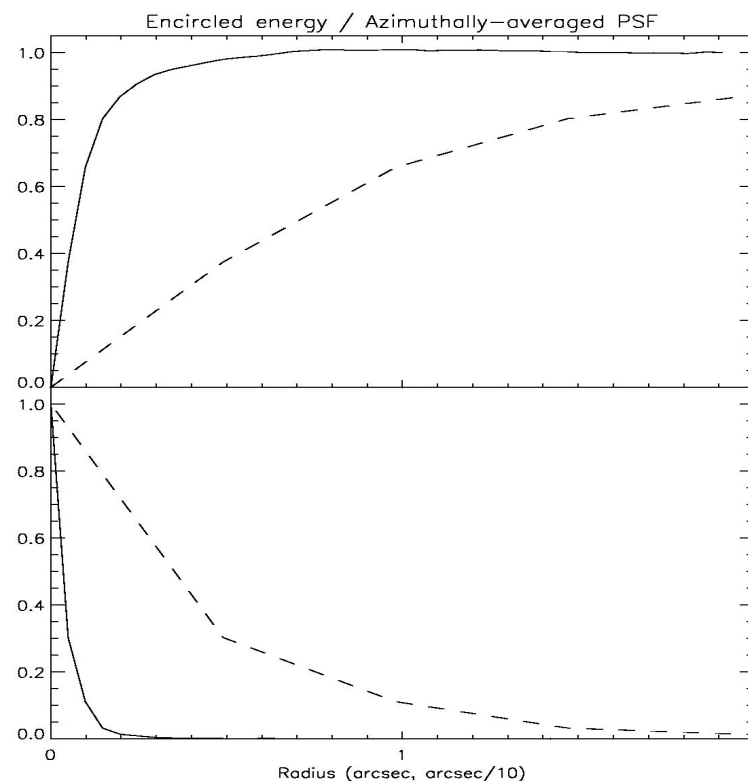
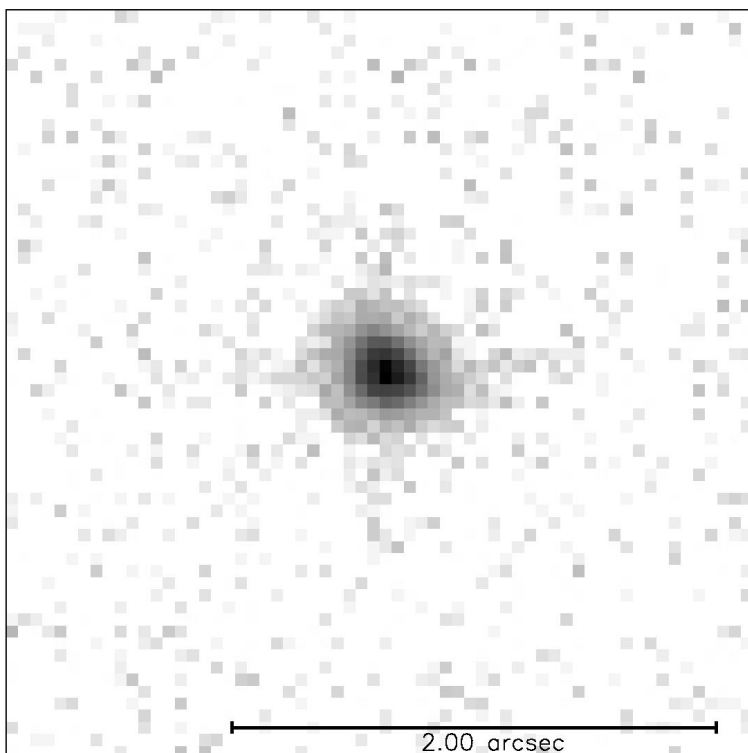
EE(d=.40): 0.8713

**Above-spec performance achieved**

EE Objective at 633 nm

(d=.25) = 67%

(d=.40) = 78%



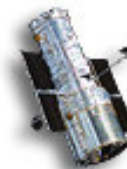
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# Principal Electronics Boxes are Complete - Several Delivered to GSFC



**IR Detector Electronics (DEB)**

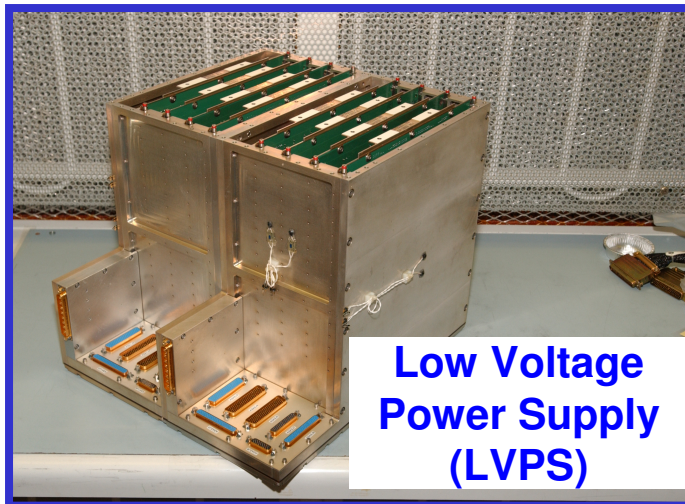


Detector e-boxes deliver  $< \text{spec read noise}$ .

**UVIS Detector Electronics (CEB)**

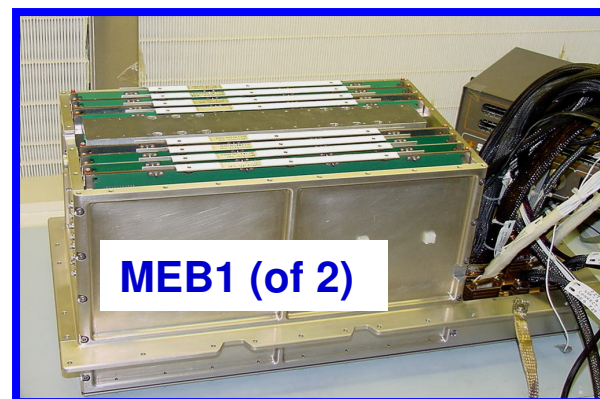


**Low Voltage Power Supply (LVPS)**



LVPS, MEBs are at GSFC for environmental testing & integration.

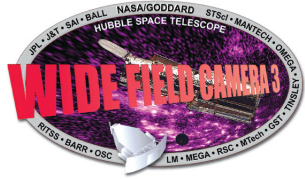
**MEB1 (of 2)**



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## TECFIRE Redesign is Complete and Fully Qualified



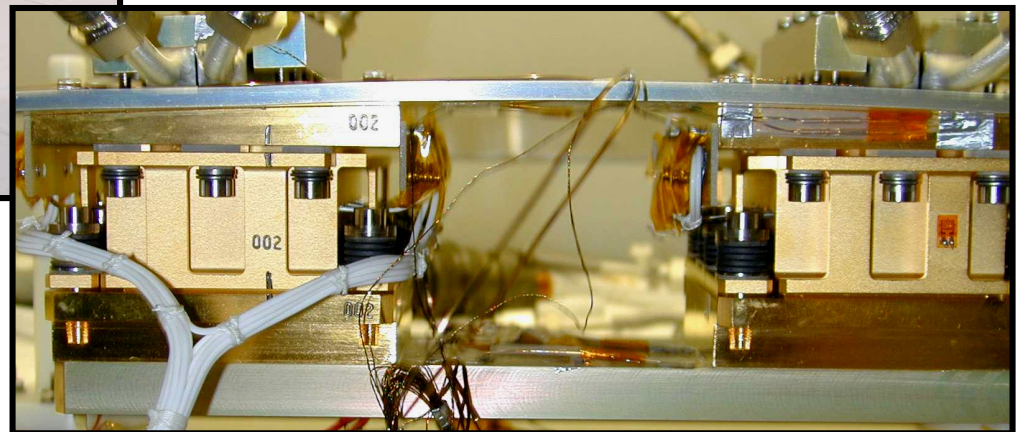
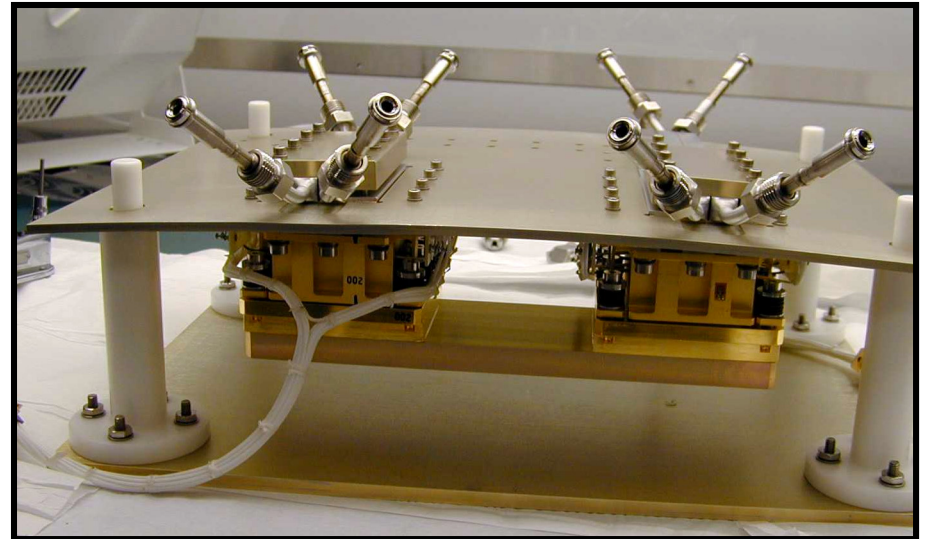
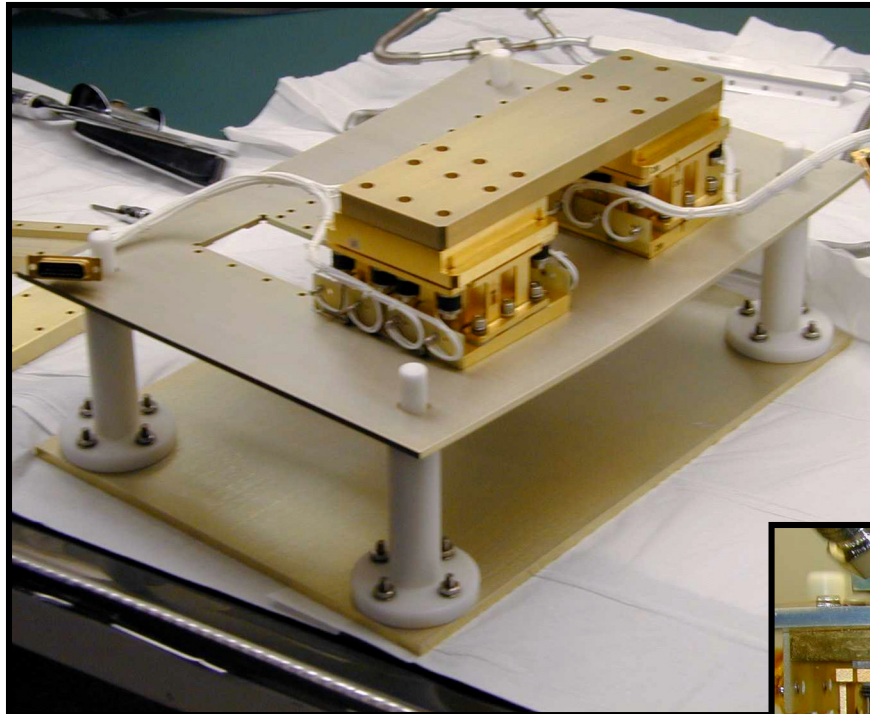
- Cooling system for IR channel – TECFIRE – (based on TECs, heat pipes, radiator) presented severe engineering challenge.
- Original design placed unacceptable mechanical loads on brittle TECs and sheared them.
- Major redesign developed incorporating gelvet (thermally conductive, mechanically compliant material).
- Through all qualification tests successfully with performance margin; tolerant to failure of single TECs.
- Gelvet successfully completed life cycle test; TECs well into similar test with no problems expected.

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# TECFIRE 'Detector Base' ETU



ETU DB TECFIRE Modules  
mounted on a simulated WFC3  
Radiator with simulated IR Flex  
Heat Pipe Saddle.

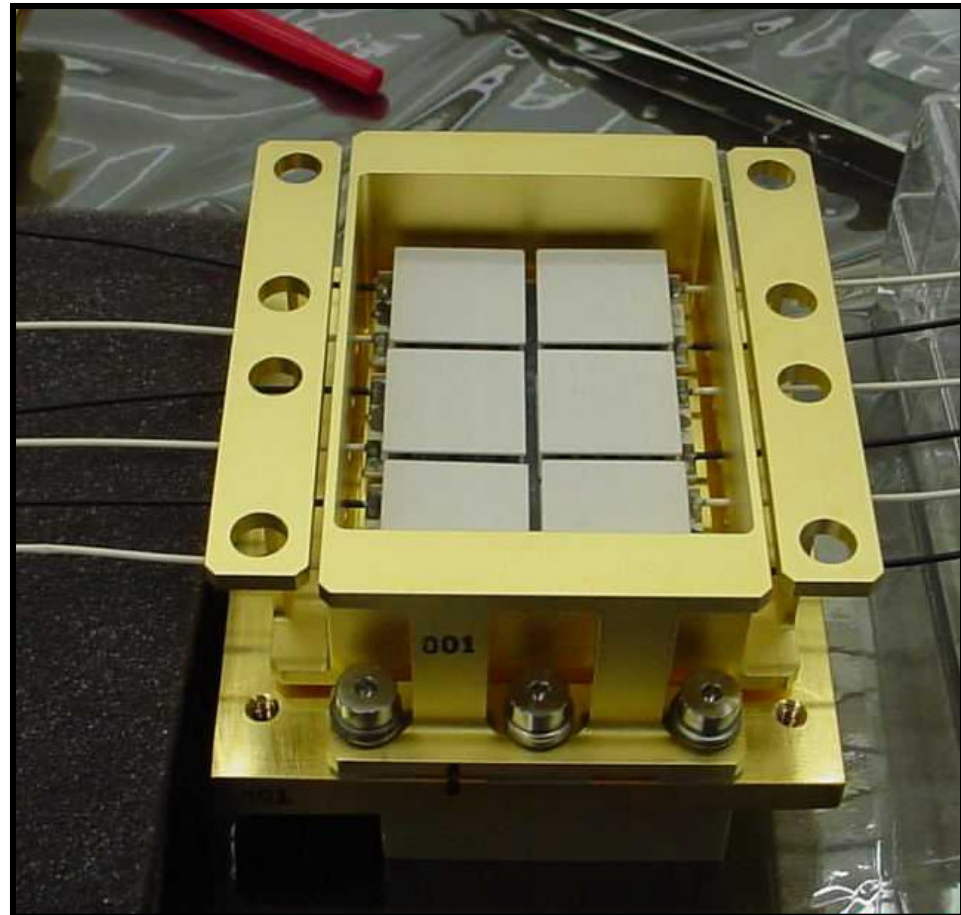
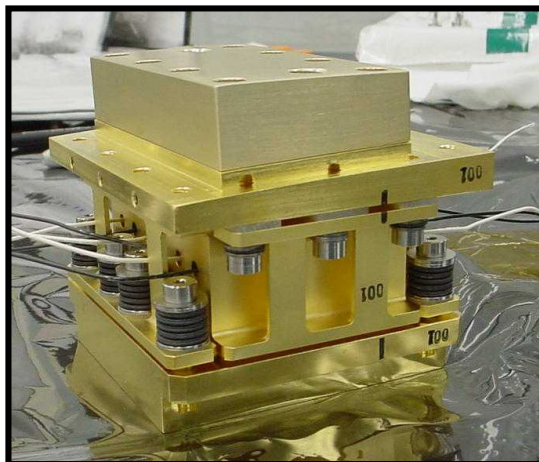
"Data contained herein is exempt from ITAR regulations under CFR 125.4(13) -- data approved for public disclosure."

Randy.A.Kimble@nasa.gov, NASA/GSFC Code 681, 301-286-5783  
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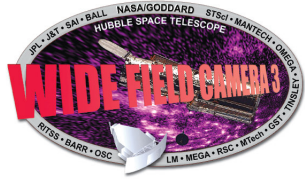


# ETU TECFIRE Single Unit Module

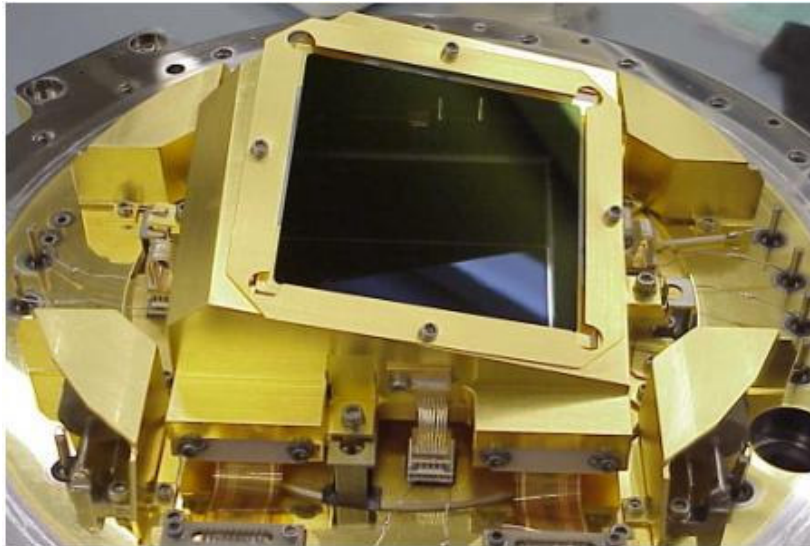


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Randy.A.Kimble@nasa.gov, NASA/GSFC Code 681, 301-286-5783  
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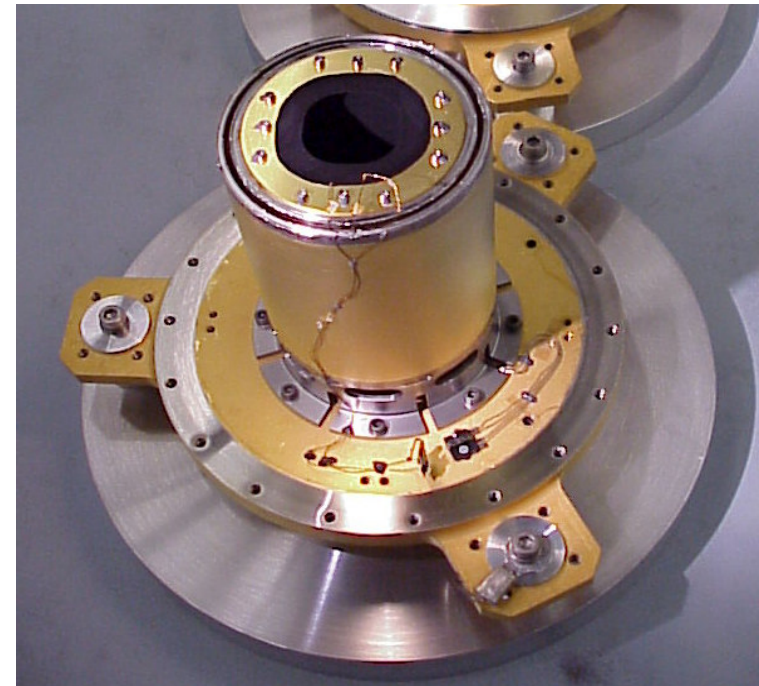


# Detector Status



- UVIS detector Build 1 is fully assembled, into environmental test.
- UVIS Build 2 is in work.

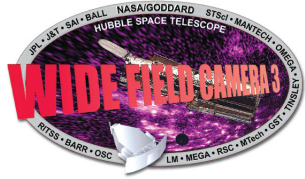
## IR Qual Unit (Vacuum Shell Removed)



- IR Qual Unit has completed thermal testing with performance margin.

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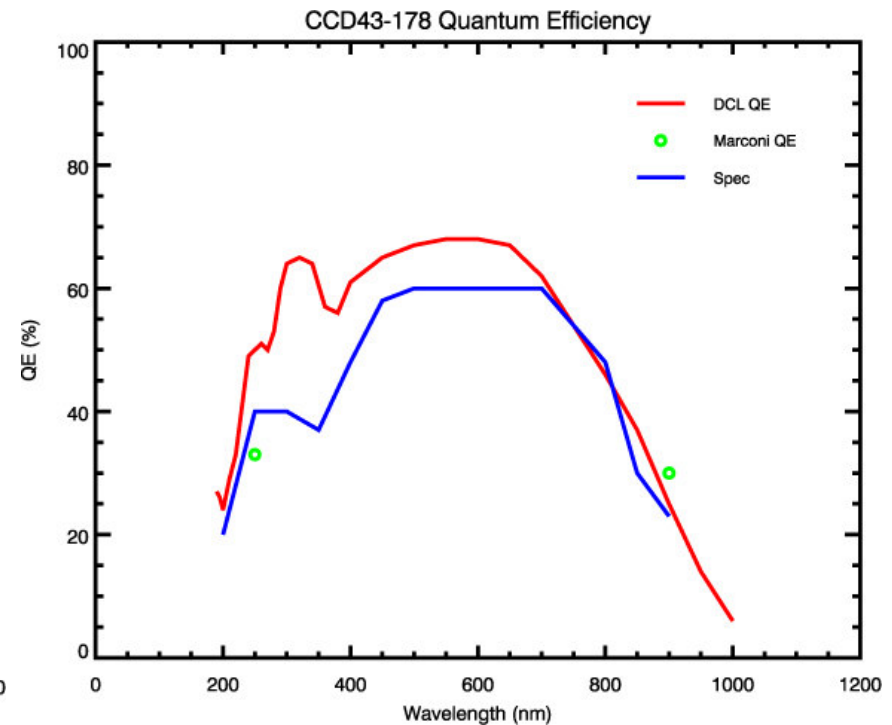
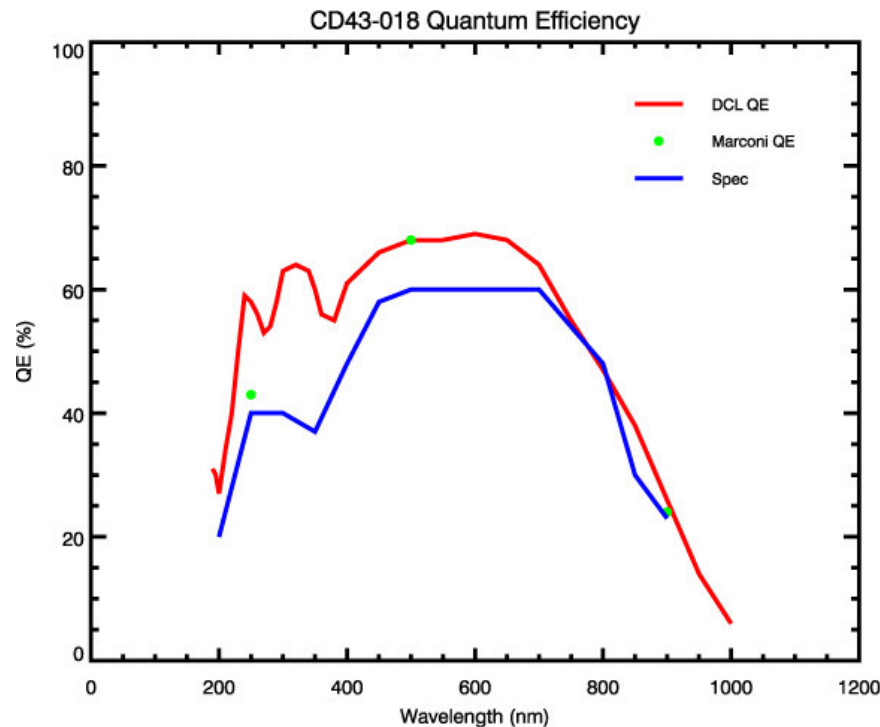
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# UVIS Build 1 Incorporates Superb Marconi CCDs



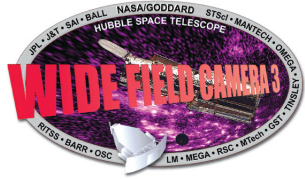
- Flight CCDs have above-spec QEs,  $<3$  e- rms read noise (3.5 e- with flight electronics), as well as excellent CTE, dark current, and cosmetics.



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# Charge Injection Selected as CTE Mitigation Strategy



- DCL has completed tests comparing charge injection and pre-flash performance for ameliorating CTE loss.
- DCL and STScI analyses concluded that charge injection offers superior performance.
- SOC concurred with recommendation to adopt charge injection for flight implementation (pre-flash hardware is still available).

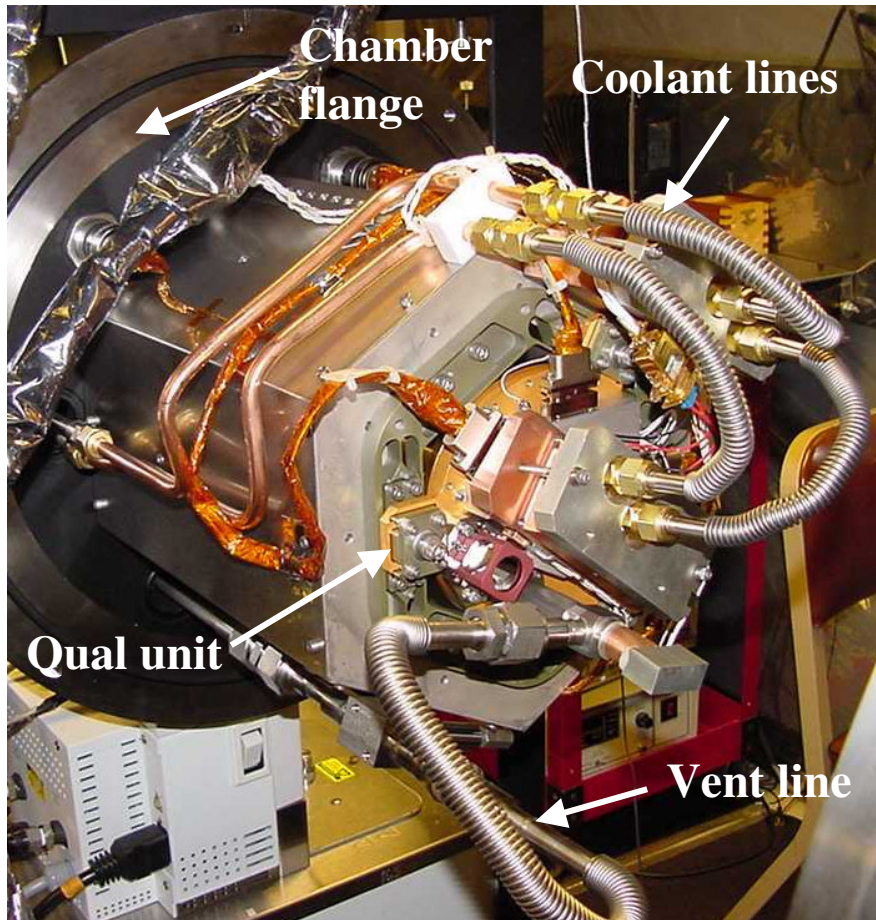
	Pre-Rad	5 Years Radiation Damage		
		Uncorrected	Charge Injection	2000e <sup>-</sup> Flat Field
CTE	0.999999	0.99984	0.999988	0.999978
<sup>55</sup> Fe Charge Loss (2048 Transfers)	0.20%	28%	2.40%	4.40%
Noise Baseline	3e <sup>-</sup>	3e <sup>-</sup>	15e <sup>-</sup>	46e <sup>-</sup>
<sup>55</sup> Fe Noise	15e <sup>-</sup>	20-100e <sup>-</sup>	20e <sup>-</sup>	53e <sup>-</sup>

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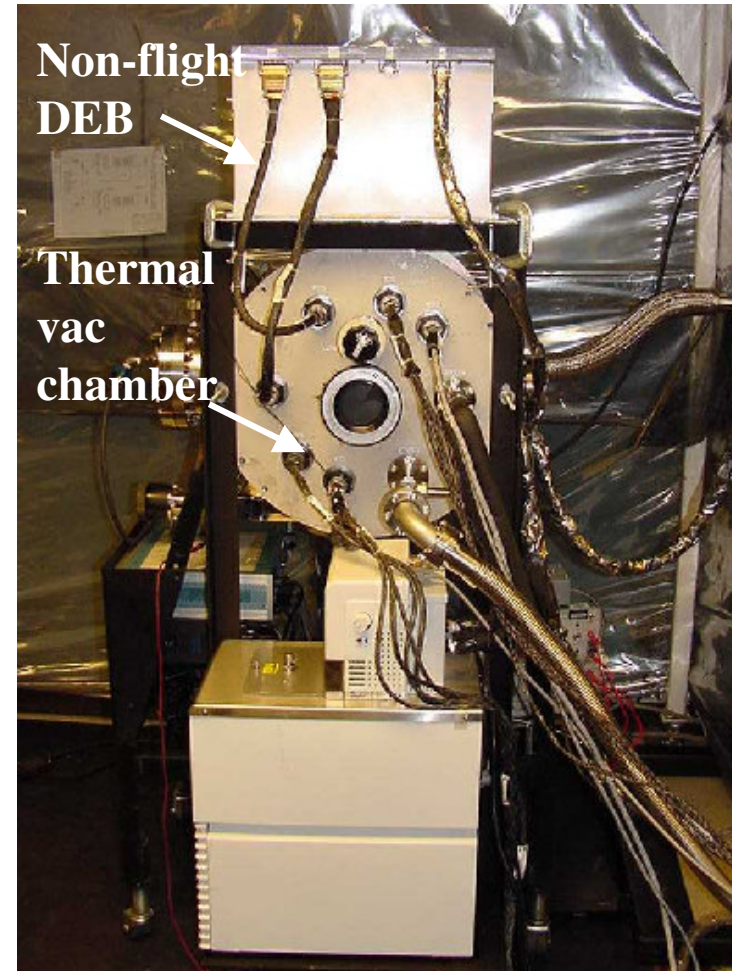
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# IR Qual Unit Successfully Completed Thermal Vacuum Testing w/Margin



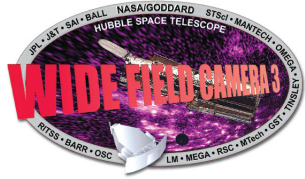
Qual unit installed on thermally-isolating structure inside thermal vac chamber



FA-2 vacuum chamber and DEB

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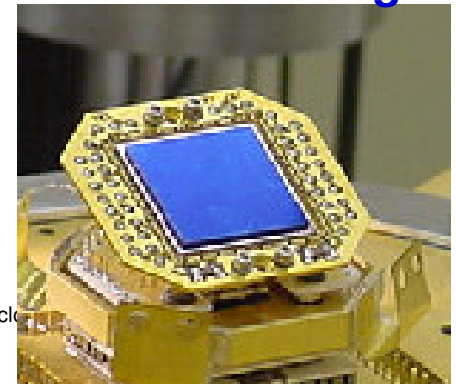


## IR Focal Plane Assembly Development Efforts are Paying Off



- Development of the short-wavelength-cutoff HgCdTe array for operation at 150K has been the biggest technical challenge of the program.
- Problems have been encountered in achieving good short wavelength QE, dark rate stability, and spec level read noise.
- Systematic development steps at Rockwell, with crucial performance feedback from DCL, have led to great progress.
- The current lot is yielding the best devices to date.
- *A flyable part, that would offer exciting performance in WFC3, is currently in hand.*
- An additional lot is still in fabrication with the potential for further improvements.

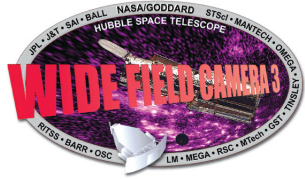
IR FPA on wedge



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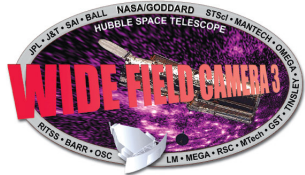
## IR FPA Problems Have Been Systematically Investigated



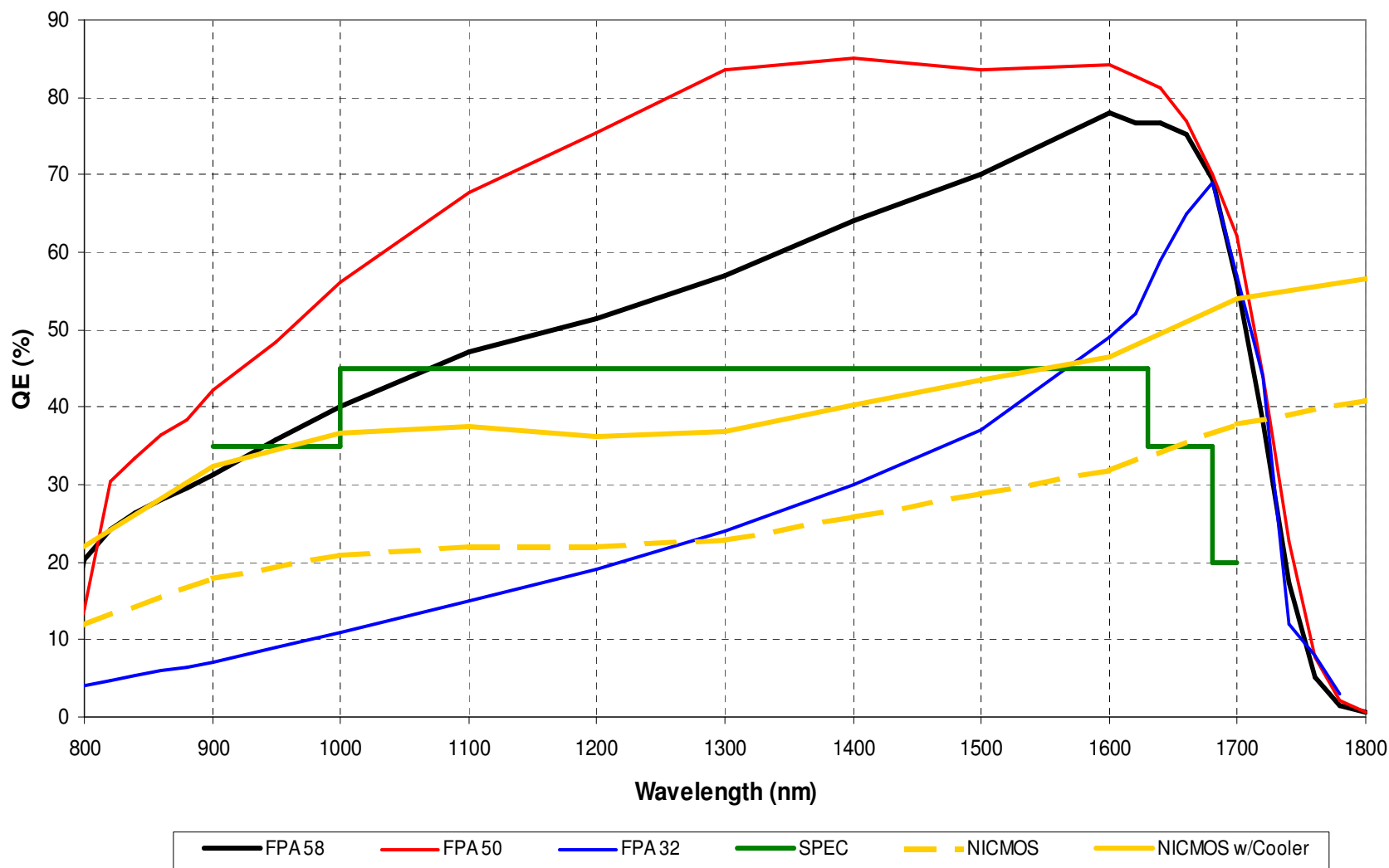
- Lot 4 parts showed severe QE shortfall at short wavelengths; below spec and below NICMOS; corrected superbly in lot 6.
- Lot 6 parts, however, showed severely unstable dark current after resets or exposure to light (some devices showed settling times of 10's of hours) – precludes straightforward on-orbit operations/calibration, though some encouraging progress has been made in developing operational strategies.
- All devices have had higher than spec readout noise (30-45 e-rms per CDS pair was typical in early lots).

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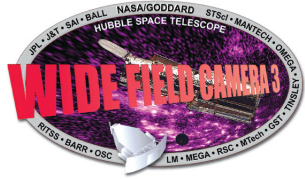
# IR Detector QE Comparison Including Best Current Candidate FPA 58



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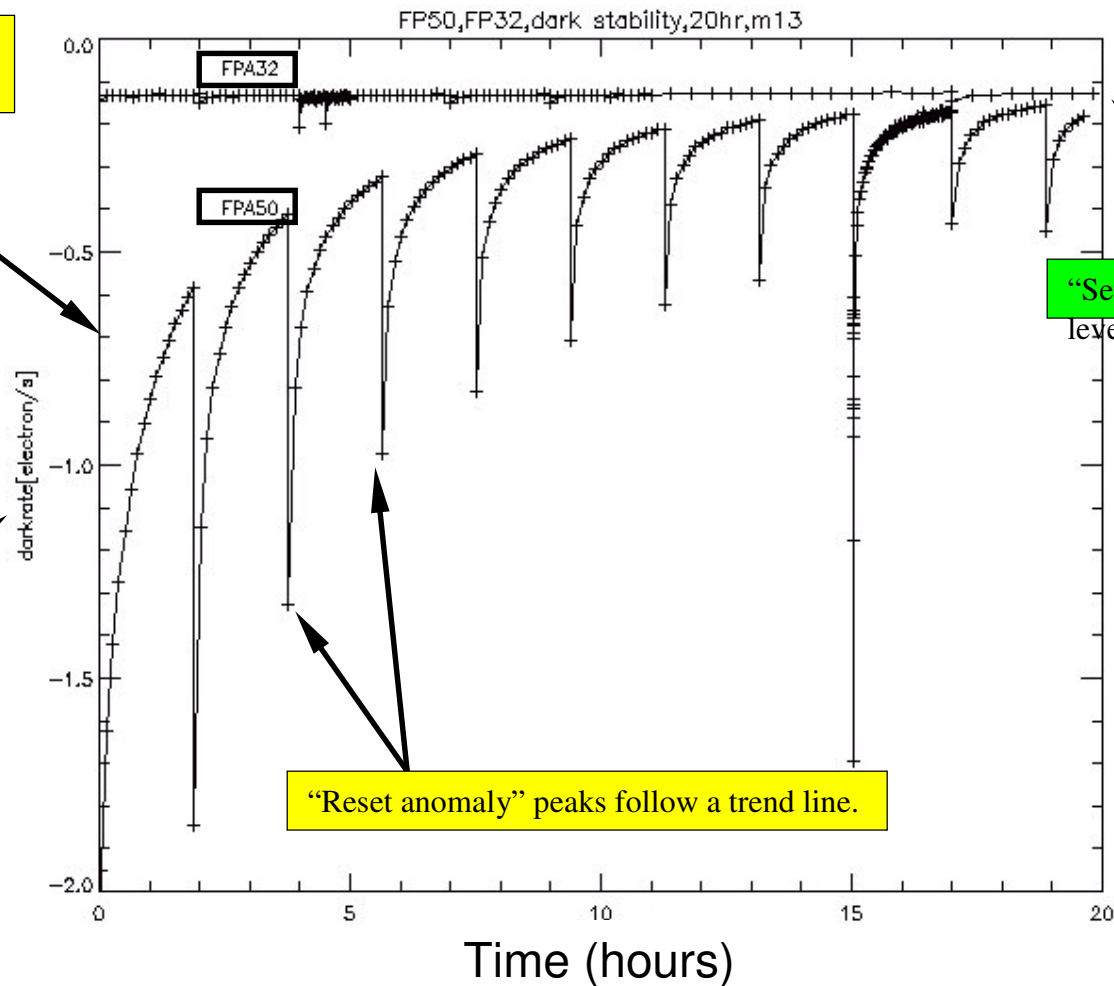


# FPA 32/50 Stability Comparison



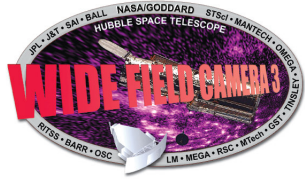
Data taken immediately after 3/4 full-well flat field.

Equivalent dark current RATE (derivative of the raw ramp signal).

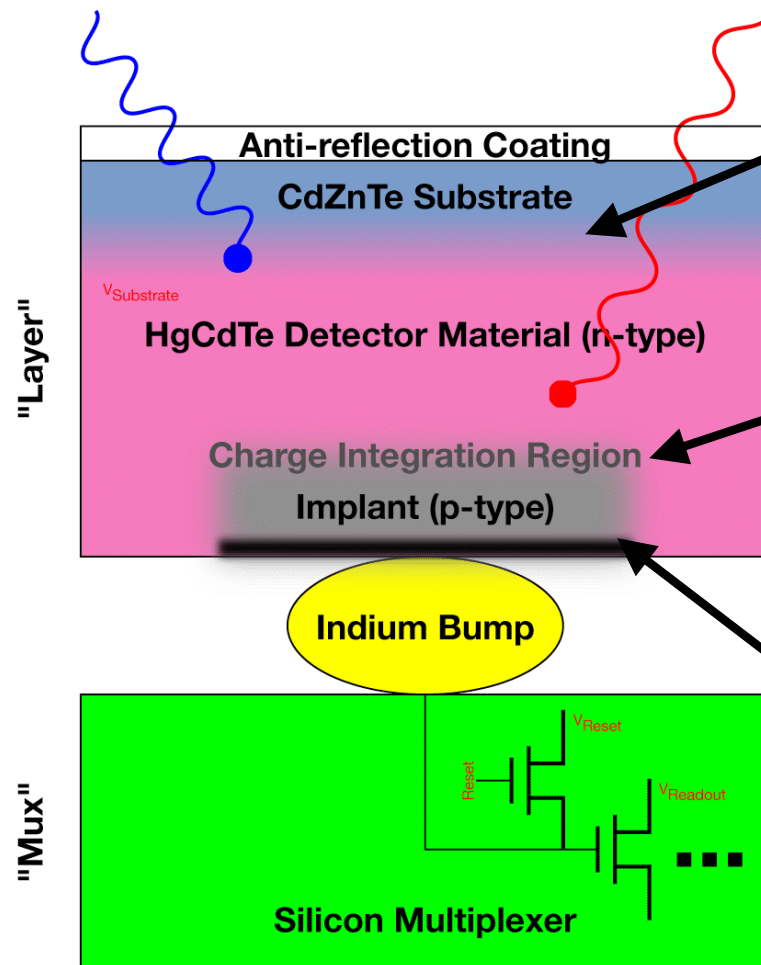


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# IR FPA Performance Issues are Physically Separable and Correctable in a Single Device



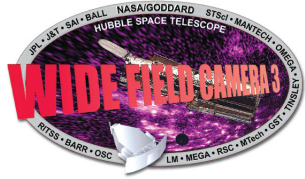
Short wavelength QE problem caused by traps in buffer region – cure with composition gradient to minimize lattice mismatches.

Instability associated with traps in charge integration region. Address with implant density/depth.

Readout noise shows temperature dependence characteristic of Johnson noise (resistive component) and 1/f drift typical of surface state trapping. Improve contact resistance and surface properties.

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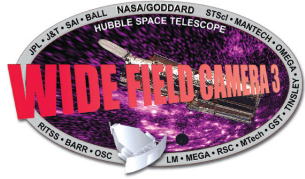
## Fabrication Status at Rockwell



- Lot 7 initiated to combine lot 4 stability approach with lot 6 QE, along with small parameter tweaks to improve read noise.
- 1<sup>st</sup> split (lot 7A) validated QE process and yielded best cosmetics and hybridization to date, but exhibited anomalously high dark current.
- Rockwell introduced IR&D lot at company expense to resolve dark current problem and optimize parameter mix for WFC3 lot 7B.
- IR&D wafers are currently in test at RSC and DCL – best devices to date for combination of QE, dark rate, stability, and noise.
- Lot 7B layers are grown; will be processed with best recipe determined from IR&D lot.

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# IR&D Lot Has Yielded Flight Candidate Device FPA 58

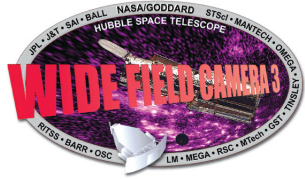


- FPA 58 is clearly the best IR FPA developed so far for WFC3.
  - QE is above spec everywhere longward of 1.05 microns.
  - Mean dark rate is 0.11 e-/sec vs. spec of 0.4 e-/sec.
  - Stability appears to be as good as best lot 4 parts.
  - Read noise is 23.5 e- rms per CDS pair and averages down further when read up the ramp.
  - 2 or 3 additional devices can be packaged from the same wafer.
- FPA 58 could be flown on WFC3 and offer exciting performance, though we of course would prefer to achieve full spec level performance with lot 7B.
- Devices from IR&D or lot 7B can be characterized by DCL and delivered to Ball for buildup of housing with no impact to I&T schedule.

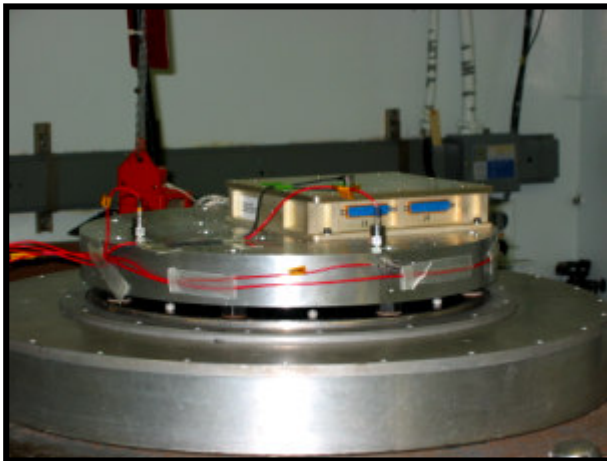
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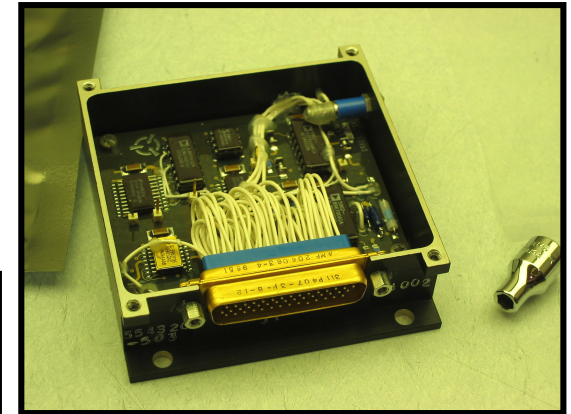




# GSFC Is Ready for WFC3 I&T



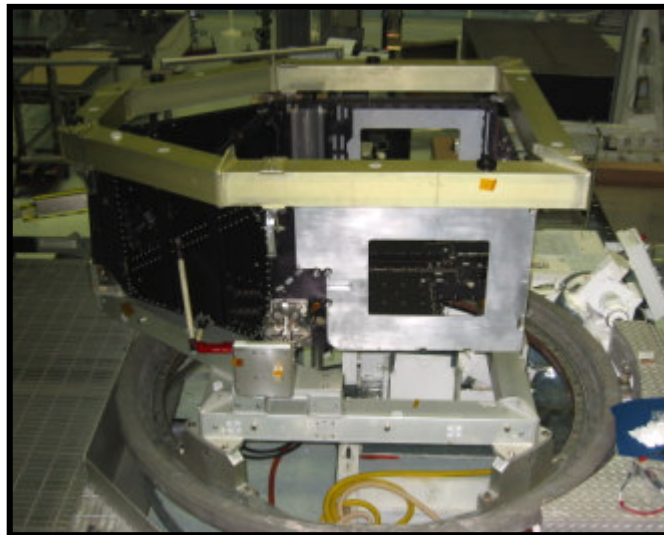
Filter Box Vibe



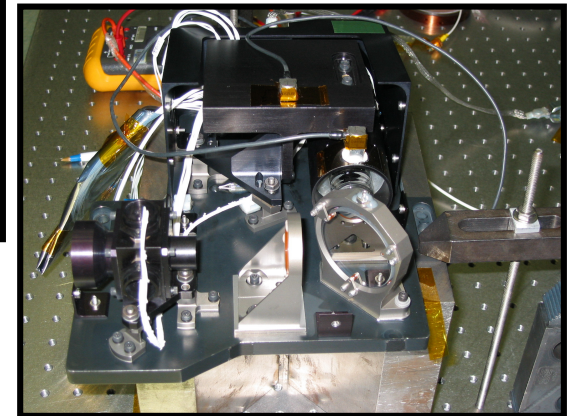
Heater Controller



TECFIRE Fit-Check



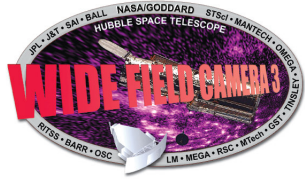
Enclosure ready for I&T



Calibration Source  
Alignment

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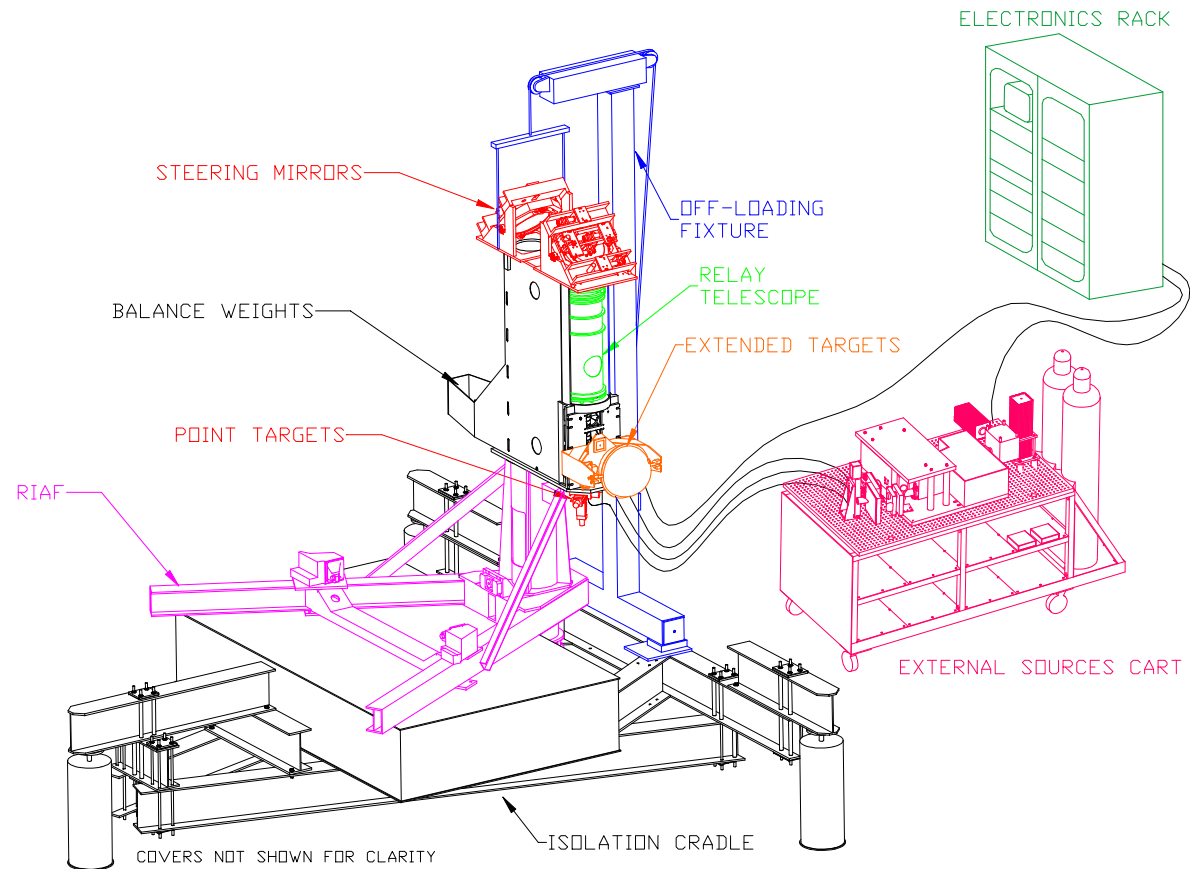
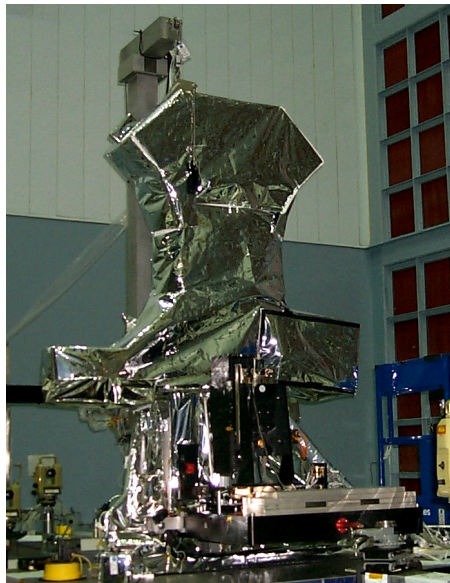
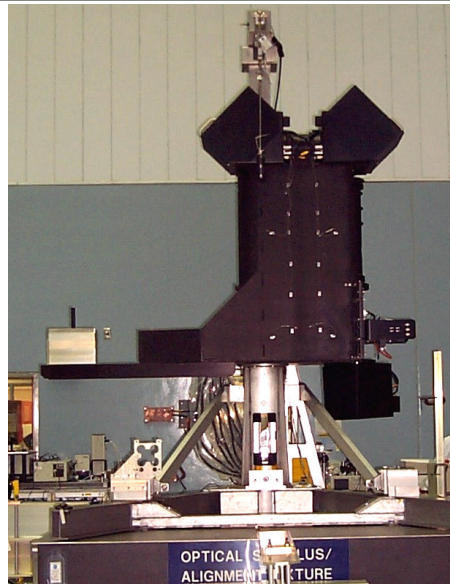
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# Optical Stimulus Has Successfully Completed Thermal-Vac, Is Ready for I&T

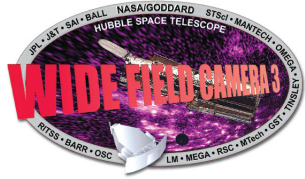


## Optical Stimulus / CASTLE

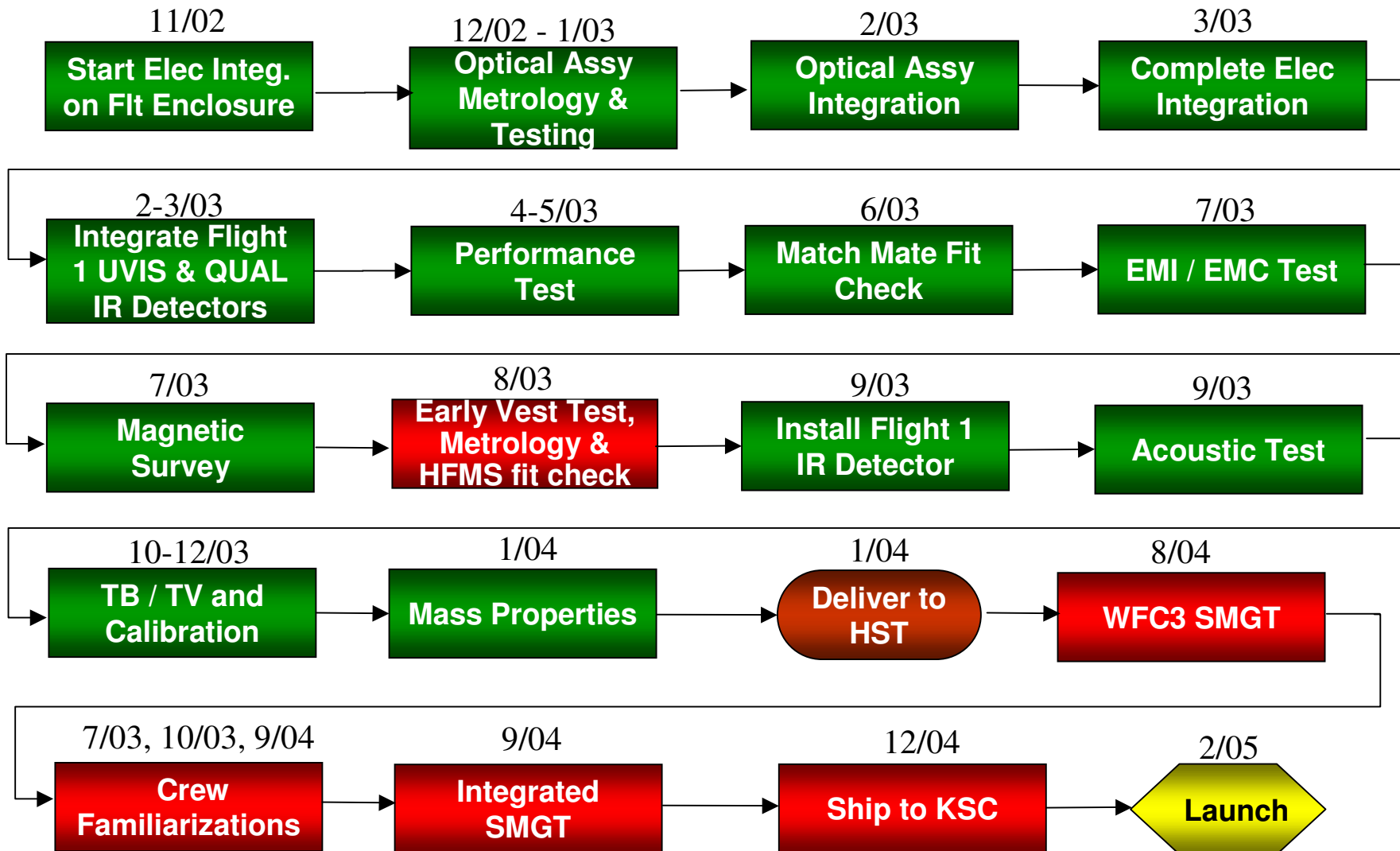


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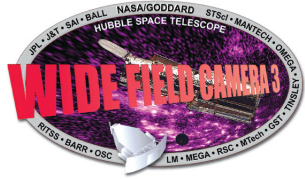
# WFC3 I&T Flow at GSFC



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# Summary



- Nearly all subsystems have been completed.
- WFC3 program is poised to begin a thorough integration and test period at Goddard.
- We are eager to deliver a superb panchromatic camera to HST in Servicing Mission 4.

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